

## Erratum: "Potential Load Reduction Using Airfoils with Variable Trailing Edge Geometry" [Journal of Solar Energy Engineering, 2005, 127(4), pp. 503–516]

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Figures 5–7 were not correct in the published article. See below for the correct figures.

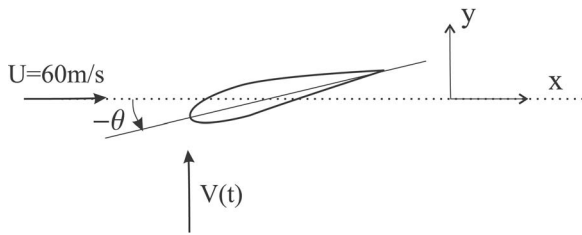


Fig. 5 Inflow setup. The  $y=0$  plane corresponds to the rotor plane.  $U$  corresponds to the rotational velocity, while  $V(t)$  corresponds to the free-stream velocity.

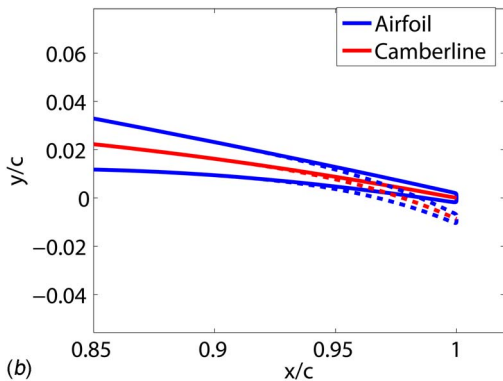
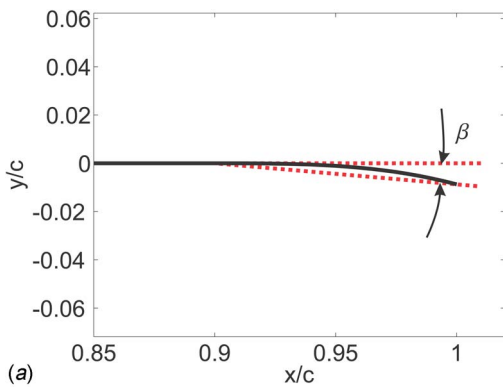


Fig. 6 Deflection corresponding to  $\beta=5$  deg. (a) Raw deflection shape. (b) Airfoil trailing edge region, with thickness.

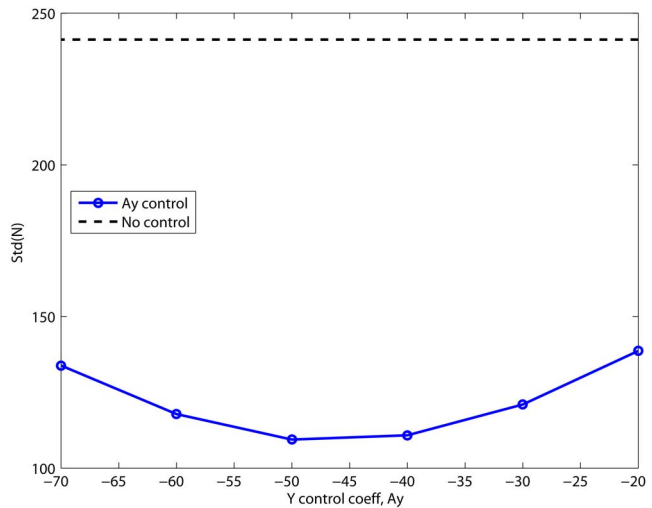


Fig. 7 Wind step response.  $\text{Std}(N)$  as a function of control parameter  $A_y$ .